Work Order ID 85	53′	77
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\*85377\*

Page 1

June-07-12 9:19	9:46 AM		$\alpha$ :							rage i
Item ID: Revision ID: Item Name:	D412-698-028  Bracket Assembly, RH (Aluminum	Doors)	Accept	*N900	040	100	)* s	etup Stari	I VI	S1*
	07/06/2012 Start Qty: 2.00	*2*		Cust Item   Customer:	ID:				· IN	S2*
Approvals:	Process Plan: MLJ QC:				ate:		R	un Start Stop	1/1	R1* R2*
Sequence ID/ Work Center II	<u> </u>		Set Up/ Run Hours	Tool ID	Tool #	Plan Code	Accept Qty	Reject Qty	Reject Number	Insp. Stamp
Draw Nbr	Revision Nbr									
DSI 9484	C		· · · · · · · · · · · · · · · · · · ·	·						
*100 *100* DC Document Control	Memo Photocop CHG003	y bluefile &type labels per	0.00 0.00 PPP D412-698-028/ DSI948	4 9 3 5	alals		79	Be	MLJ G	12-6-2
110	Pick Kit		0.00	•						
*110* Packaging Packaging	Memo		0.00				2			12/04/27
120	QC4- 100% Inspect ki	ts for completeness	0.00	`						
*120* QC Quality Control	Memo		0.00	13/13			CHU			

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	WORK ORDER CHANGES							
STEP	PROCEDURE CHANGE		Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Appróval QC Inspector	
				•			. !	
}								
_	STEP	PROCEDURE CHANGE	PROCEDURE CHANGE	PROCEDURE CHANGE By	PROCEDURE CHANGE By Date	PROCEDURE CHANGE By Date Qty	PROCEDURE CHANGE By Date Qty Chief Eng / Prod Mgr	

Part No: _	PAR #:	Fault Category:	NCR: Yes No DQA:	Date:
	Resolution:	Disposition:	QA: N/C Closed:	Date:
		MORK ORDER NON-CONFORMA	NCE (NCB)	

NCR:			WORK ORD	ER NON-CONFORMANC	E (NCH)			•	
	T	Description of NC	Description of NC Corrective Action Section B						
DATE	STEP	Section A	Initial Chief Eng	Action Description Chief Eng	Sign & Date	Verification Section C	Approval Chief Eng	Approval QC Inspector	
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W	ork	Ord	ler ID	85377
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Page 2

June-07-12 9:19:46 AM D412-698-028 Item ID: Accept \*N900040100\* Setup Start **Revision ID:** Item Name: Bracket Assembly, RH (Aluminum Doors) **Start Date:** 07/06/2012 Start Qty: 2.00 **Cust Item ID: Required Date:** 21/06/2012 Req'd Qty: 2.00 **Customer:** Reference: Run Process Plan: Approvals: Date:\_\_\_\_ Tooling: Date: Stop QC: Date: SPC (Y/N): Date: Sequence ID/ Operation Set Up/ Tool ID Tool # Plan Reject Accept Reject Insp. **Work Center ID** Description **Run Hours** Qty Code Qty Number Stamp 130 0.00 \*130\* Packaging 0.00 Memo Packaging Identify and pack for shipping as per PPP D412-698-028/ DSI 9484 Location: 140 QC21- Final Inspection - Work Order Release 0.00 \*140\* QC 0.00 Memo Quality Control

	Dart	Aero	space	: Ltd
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W/O:		WORK ORDER CHANGES											
DATE	STEP	PROCEDURE CHANGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Appróval QC Inspector						
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Part No	:	PAR #: Fault Category:	NCR: Yes	No <b>DQ</b>	A:	_ Date: _							

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### **Picklist Print**

June-07-12 9:19:52 AM

Work Order ID: 85377

\*85377\*

Parent Item:

D412-698-028

\*D412-698-028\*

Parent Item Name: Bracket Assembly, RH (Aluminum Doors)

Start Date: 07/06/2012

**Required Date: 21/06/2012** 

Page 1

Start Qty: 2.00

Required Qty: 2.00

Comments:

IPP RevA: new issue DD 09.11.30 verified by:EC

IPP Rev:B

11.03.31 AS PER ECN11-531 verified by:EC

Component Item ID/ Item Name	Replacement Item ID	Mfg/ Purch	Bin Item	Primary Location	Last Location	Route Seq ID	Unit of Measure	Qty on Hand	Qty per Kit	Total Qty	Qty Issued	Date Issued	Status
D3199-4		Manufactured	No			110	Each	43.0000	4.0	<b>7</b> 2	<i>i</i> /		
*D84100_1 *	•								**	, –	1 4	<b>On</b>	



<b>Location</b>	Loc Qty	Loc Code
ST236B	43	
77735	43	

Dart Aerospace	Ltd
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W/O:		WORK ORDER CHANGES											
DATE	STEP	PROCEDURE CHANGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector						

Part No: *	PAR #:	Fault Category:	NCR: Yes No	DQA:	Date:	
Resolution:		Disposition:	QA: N/C Closed	<b>i</b> :	Date:	

NCR:		WORK ORDER NON-CONFORMANCE (NCR)								
		Description of NC		Corrective Action Section B		Verification	Approval	Approval		
DATE	STEP	Section A	Initial Chief Eng	Action Description Chief Eng	Sign & Date	Section C	Chief Eng	QC Inspecto		
		•								
							-			

TO AMEND INSTALLATION INSTRUCTIONS IIN-D412-698 REV. E OR LATER APPROVED REVISION AND ICA-D412-698 REV. 5 OR LATER APPROVED REVISION

REF. TCCA STC: SH92-43 REF. FAA STC: SR01446NY **REF. EASA.IM.R.S.00709** 

Field experience has shown that the forward D3199-1 brackets that are part of the D412-698-011/-012/-017 Cabin Door Roller Kits and forward D3675-1 brackets that are part of the D412-698-031/-032/-033 Cabin Door Roller Kits may wear prematurely if the aircraft is flown a significant amount of time with the sliding doors open. Dart has developed the D412-698-027/-028 kits that includes a longer D3199-3/-4 bracket for installation in the first forward position of the D412-698-011/-012/-017 kits to reduce wear on these forward brackets on aircraft equipped with aluminum doors. Likewise, Dart has developed the D412-698-037/-038 kits that includes a longer D3675-5/-6 bracket for installation in the first forward position of the D412-698-031/-032/-033 kits to reduce wear on these forward brackets on aircraft equipped with composite doors.

The D412-698-027/-028/-037/-038 kits may be installed as follows:

1) Remove the sliding door from the aircraft per Section 52.1 of ICA-D412-698

2) Remove the first forward D3199-1 or D3675-1 brackets as applicable.

3) Install the D3199-3/-4 or D3675-5/-6 bracket as applicable in lieu of the removed -1 bracket. Refer to Figure 1 Sheet 2 (aluminum doors) or Figure 2 Sheet 3 (composite doors).

4) Torque hardware per ICA-D412-698.

5) Re-install sliding door on aircraft per Section 52.2/52.3 of ICA-D412-698 as applicable.

6) Make entry in technical records of aircraft indicating that DSI 9484 has been completed.

### PARTS LIST

QTY -027	QTY. -028	QTY. -037	QTY. -038	PART NUMBER	DESCRIPTION
X	X		D412-698-027	BRACKET KIT, LH (ALUMINUM DOORS)	
	Х			D412-698-028	BRACKET KIT, RH (ALUMINUM DOORS)
		Х		D412-698-037	BRACKET KIT, LH (COMPOSITE DOORS)
			Х	D412-698-038	BRACKET KIT, RH (COMPOSITE DOORS)
1				D3199-3	FORWARD BRACKET, LH
	1			D3199-4	FORWARD BRACKET, RH
		1		D3675-5	FORWARD BRACKET, LH
			1	D3675-6	FORWARD BRACKET, RH

### WEIGHT AND BALANCE

Installation	Weight	ht Lateral		Long	itudinal	Applicability	
		Arm	Moment	Arm	Moment	- Approcability	
D412-698-027	0.16 lb	-48.0 in	-7.68 in-lb	102.3 in	16.37 in-lb	Aircraft Equipped with	
Bracket Kit	0.07 kg	-1.22 m	-0.09 m-kg	2.60 m	0.18 m-kg		
D412-698-028	0.16 ibs	+48.0	+7.68 in-lb	102.3 in	16.37 in-lb	P/N 205-032-669-XXX	
Bracket Kit	0.07 kg	+1.22 m	+0.09 m-kg	2.60 m	0.18 m-kg	Aluminum Doors	
D412-698-037	0.18 lb	-48.0 iп	-8.64 in-lb	102.3 in	18.41 in-lb	Aircraft Equipped with	
Bracket Kit	0.08 kg	-1.22 m	-0.10 m-kg	2.60 m	0.21 m-kg		
D412-698-038	0.18 lb	+48.0 in	+8.64 in-lb	102.3 in	18.41 in-lb	P/N 412-030-300-XX	
Bracket Kit	0.08 kg	+1.22 m	+0.10 m-kg	2.60 m	0.21 m-kg	Composite Doors	

CANADA DEPARTMENT OF TRANSPORT AIRCRAFT CERTIFICATION **BRANCH** DAO # 01-O-01

APPROVED D. SHEPHERD (DE # 02)

DATE: CERT. NO.: ISSUE NO .:

11.03.21 SH92-43

DESIGN			
REV.	DESCRIPTION	BY	DATE
Α	NEW ISSUE	85	09.10.16
В	ADDED -037/-038 KIT FOR COMPOSITE DOORS; UPDATED TITLE ACCORDINGLY.	МВ	10.12.17
С	UPDATED -027/-028/-037/-038 KITS AND INSTALLATION	MB	11.03.21

DEGIGIA	94	HAWKESBURY, ONTARIO, CANADA				
DRAWN	A					
CHECKED	ASS	DRAWING NO.	REV. C			
MFG. APPR.	N/A	DSI 9484 SHEE	T 1 OF 3			
APPROVED	N	TITLE	SCALE			
DE APPR.	-4	D412-698-027/-028/-037/-038 Kits	NTS			
DATE 11	13 21	COPYRIGHT © 2009 BY DART AEROSPACE LTD. THIS DOCUMENT IS PRIVATE AND CONFIDENTIAL AND IS SUPPLED ON THE EXPRESS CONDITION THAT IT IS				

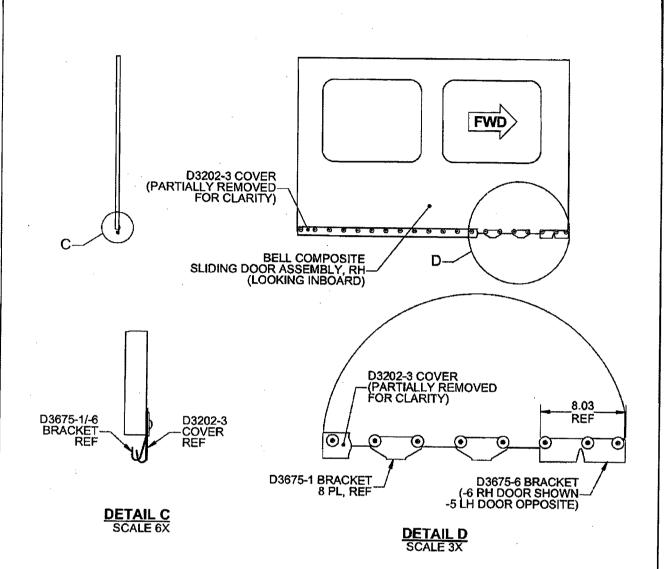
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WITHOUT NOTICE

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# FIGURE 2: D3675-5/-6 BRACKET INSTALLATION ON COMPOSITE DOORS (-6 SHOWN, -5 OPPOSITE)



DESIGN	<i>8</i> €}	DART AEROSPACE LTD HAWKESBURY, ONTARIO, CANADA				
DRAWN	13					
CHECKED	TASS	DRAWING NO.	REV. C			
MFG. APPR.	N/A)	DSI 9484 SHEET	3 OF 3			
APPROVED	1 /VP		SCALE			
DE APPR.	<b>-</b> ##-	D412-698-027/-028/-037/-038 Kits	NTS			
DATE 11.0	3.21	COPYRIGHT © 2009 BY DART AEROSPACE LTD THIS DOCUMENT BY PRIVATE AND CONFIDENTIAL AND IS SUPPLIED ON THE EXPRESS CONSTITUTION THAT IT IS NOT TO BE USED FOR ANY PAPTOSE OR OF COPED ON COMMANDATION TO ANY OTHER PERSON WITHOUT				

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### DART SERVICE INSTRUCTION

TO AMEND INSTALLATION INSTRUCTIONS IIN-D412-698 REV. E OR LATER APPROVED REVISION AND ICA-D412-698 REV. 5 OR LATER APPROVED REVISION
REF. TCCA STC: SH92-43
REF. FAA STC: SR01446NY
REF. EASA.IM.R.S.00709

Field experience has shown that the forward D3199-1 brackets that are part of the D412-698-011/-012/-017 Cabin Door Roller Kits and forward D3675-1 brackets that are part of the D412-698-031/-032/-033 Cabin Door Roller Kits may wear prematurely if the aircraft is flown a significant amount of time with the sliding doors open. Dart has developed the D412-698-027/-028 kits that includes a longer D3199-3/-4 bracket for installation in the first forward position of the D412-698-011/-012/-017 kits to reduce wear on these forward brackets on aircraft equipped with aluminum doors. Likewise, Dart has developed the D412-698-037/-038 kits that includes a longer D3675-5/-6 bracket for installation in the first forward position of the D412-698-031/-032/-033 kits to reduce wear on these forward brackets on aircraft equipped with composite doors.

The D412-698-027/-028/-037/-038 kits may be installed as follows:
 Remove the sliding door from the aircraft per Section 52.1 of ICA-D412-698
 Remove the first forward D3199-1 or D3675-1 brackets as applicable.
 Install the D3199-3/-4 or D3675-5/-6 bracket as applicable in lieu of the removed -1 bracket. Refer to Figure 1 Sheet 2 (aluminum doors) or Figure 2 Sheet 3 (composite doors).
 Torque hardware per ICA-D412-698.
 Re-install sliding door on aircraft per Section 52.2/52.3 of ICA-D412-698 as applicable.
 Make entry in technical records of aircraft indicating that DSI 9484 has been completed.

### **PARTS LIST**

QTY -027	QTY. -028	QTY. -037	QTY. -038	PART NUMBER	DESCRIPTION
X				D412-698-027	BRACKET KIT, LH (ALUMINUM DOORS)
	X	i		D412-698-028	BRACKET KIT, RH (ALUMINUM DOORS)
······································		X		D412-698-037	BRACKET KIT, LH (COMPOSITE DOORS)
			X	D412-698-038	BRACKET KIT, RH (COMPOSITE DOORS)
1				D3199-3	FORWARD BRACKET, LH
	Lan 1 1	F		D3199-4	FORWARD BRACKET, RH
	1	1		D3675-5	FORWARD BRACKET, LH
	<del>                                     </del>		1	D3675-6	FORWARD BRACKET, RH

### WEIGHT AND BALANCE

Installation	Weight	La	teral	Long	itudinal	Applicability
Matanation	rreigni	Arm	Moment	Arm	Moment	
D412-698-027	0.16 lb	-48.0 in	-7.68 in-lb	102.3 in	16.37 in-lb	Aircraft Equipped with P/N 205-032-669-XXX
Bracket Kit	0.07 kg	-1.22 m	-0.09 m-kg	2.60 m	0.18 m-kg	
D412-698-028	0.16 lbs	+48.0	+7.68 in-lb	102.3 in	16.37 in-lb	Aluminum Doors
Bracket Kit	0.07 kg	+1.22 m	+0.09 m-kg	2.60 m	0.18 m-kg	
D412-698-037	0.18 lb	-48.0 in	-8.64 in-lb	102.3 in	18.41 in-lb	Aircraft Equipped with P/N 412-030-300-XXX
Bracket Kit	0.08 kg	-1.22 m	-0.10 m-kg	2.60 m	0.21 m-kg	
D412-698-038	0.18 lb	+48.0 in	+8.64 in-lb	102.3 in	18.41 in-lb	Composite Doors
Bracket Kit	0.08 kg	+1.22 m	+0.10 m-kg	2.60 m	0.21 m-kg	

CANADA DEPARTMENT OF TRANSPORT AIRCRAFT CERTIFICATION BRANCH DAO # 01-0-01

PPROVED D. SHEPHERD (DE # 02)

CERT. NO.: SH92-43 ISSUE NO.:

C	UPDATI	ED -027/-028/-0	MB	11.03.21				
В	ADDED TITLE A	-037/-038 KIT I	МВ	10.12.17				
Α	NEW IS	SUE		85	09.10.16			
REV.			DESCRIPTION	BY	DATE			
DESIG	N	<b>SE</b>	DART AEROSPA	CEL	TD .			
DRAW	DRAWN A		HAWKESBURY, ONTARIO, CANADA					
CHECK	ŒD	ASS	DRAWING NO.		REV. C			
MFG. A	PPR.	N/A	DSI 9484		SHEET 1 OF 3			
APPRO	VED	NU	TITLE	,	SCALE			
DE APP	PR.	-	D412-698-027/-028/-037	/-038 I	Kits NTS			
DATE	11.0	3.21	COPYRIGHT @ 2009 BY DART AEROSPACE LTD.  THIS DOCUMENT IS PRIVATE AND COMPORTINA AND IS SIDE BOO WITE EXPRESS CONCINION THAT IT IS  NOT TO BE USED FOR ANY FURPHES OR COPYED OR COMPLEADURIED TO ANY OTHER PETSON WITHOUT  TO BE USED FOR ANY FURPHES OR COPYED OR COMPLEADURIED TO ANY OTHER PETSON WITHOUT  WITHOUT ANY FURPHES OR THE PETSON WITHOUT THE ABORDANCE IN THE ABORDANCE IN THE PETSON WITHOUT  THE PETSON WITHOUT THE PETSON WITHOUT THE ABORDANCE IN THE ABORDANC					

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QC4-100% Inspect kits for completeness

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QC

Memo

Quality Control

Dart Aerospace Ltd
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W/O:			WO	RK ORDER CHANGES	3			-
DATE	STEP	PRO	OCEDURE CHA	NGE	Ву	Date Qt	Approval Chief Eng / Prod Mgr	Approval QC Inspector
							, roa mg	
Part No		PAR #:	Fault Cateç	Jory:	NCR: Yes	No DQA: _	Date: _	
	R	esolution:	Disposition	n:	QA: N/C Clo	sed:	Date: _	
NCR:			WORK ORDE	R NON-CONFORMAN	CE (NCR	)		
DATE	STEP	Description of NC		Corrective Action Section		Verificatio		Approval
DAIL	O.L.	Section A	Initial Chief Eng	Action Description Chief Eng	Sign & Date	Section C	Chief Eng	QC Inspector
								}
					<u>.</u> :			

\*85380\*

Page 2

Item ID: Revision ID:	D412-698-03			Accept	*N900	040	1100	ገ*	Setup		*N	S1*
Item Name:	Bracket Assen	nbly, LH (Composite Do	oors)							Stop	*N.	S2*
Start Date:	07/06/2012	Start Qty: 1.00	*1*		Cust Item 1	ID:						
Required Date:	21/06/2012	Req'd Qty: 1.00	*1*		Customer:							
Reference:			•									
Approvals:	Process Pla	n:	Date:	Tooling:	D:	ate:	-		Run	Start	*N	R1*
	QC:		Date:	_ SPC (Y/N):	Da	ate:				Stop	*N	R2*
Sequence ID/ Work Center II	D	Operation Description	·	Set Up/ Run Hours	Tool ID	Tool #	Plan Code	Accep Qty	ot Rej Qty		Reject Number	Insp. Stamp
130				0.00								_
*130* Packaging		Memo		0.00								-80
Packaging			pack for shipping as per	PPP D412-698-037/ DSI 9	484							12-6-
140		QC21- Final Inspection -	Work Order Release	0.00					i	_ 1	1.0	$\gamma$
*1 <b>4</b> 0*									/	2/1	14	*/
QC		Memo		0.00					<u>.</u>		<u> </u>	
Quality Control											M	5070°

### **Dart Aerospace Ltd**

W/O:			W	ORK ORDER CHANG	ES		-	·· · · · · · · · · · · · · · · · · · ·	
DATE	STEP	PRO	CEDURE CH	ANGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector
						····			
Part No	:	PAR #:	Fault Cate	egory:	NCR: Yes	No <b>DQ</b>	A:	Date:	
		esolution:							
NCR:		\	WORK ORE	DER NON-CONFORMA	NCE (NCR	)			
DATE	STEP	Description of NC Section A	Initial Chief Eng	Corrective Action Section  Action Description  Chief Eng	on B Sign & Date		cation ion C	Approval Chief Eng	Approval QC Inspector
			5	Ond Eng					
							••		
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### Picklist Print

June-07-12 9:20:40 AM

Work Order ID: 85380

\*85380\*

Parent Item:

D412-698-037

\*D412-698-037\*

Parent Item Name: Bracket Assembly, LH (Composite Doors)

**Start Date:** 07/06/2012

**Required Date: 21/06/2012** 

Start Qty: 1.00

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Required Qty: 1.00

**Comments:** 

IPP Rev:A 11.01.11 new issue DD verified by:EC 11.03.31 AS PER ECN11-531 verified by:EC

IPP Rev:B

	· ·			······································	•									
Component Item ID/	Replacement	Mfg/	Bin	Primary	Last	Route	Unit of	Qty on	Qty per Kit	Total	Qty	Date	Status	
Item Name	Item ID	Purch	Item	Location	Location	Seq ID	Measure	Hand		Qty	Issued	Issued	2111113	
	•											·		
D3675-5		Manufactured	No			110	Fach	3.0000		1				



Location	Loc Qty	Loc Code
ST236A	3	
69573	3	

### Dart Aerospace Ltd

Dail Ac	ospace	Liu							
W/O:			W	ORK ORDER CHANGE	S				•
DATE	STEP	PRO	CEDURE CHA	NGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector
								*	
i.									
Part No	:	PAR #:	Fault Cate	gory:	NCR: Yes	No <b>DQ</b>	A:	Date:	
	· R	esolution:	Dispositio	n:	QA: N/C Cl	osed:		Date: _	
NCR:		\	WORK ORD	ER NON-CONFORMA	NCE (NCR	)			
DATE	0750	Description of NC		Corrective Action Section		Verific	cation	Approval	Approval
DATE	STEP	Section A	Initial Chief Eng	Action Description Chief Eng	Sign & Date		on C	Chief Eng	QC Inspector
					į				
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TO AMEND INSTALLATION INSTRUCTIONS IIN-D412-698 REV. E OR LATER APPROVED REVISION
AND ICA-D412-698 REV. 5 OR LATER APPROVED REVISION
REF. TCCA STC: SH92-43
REF. FAA STC: SR01446NY
REF. FAA STC: SR01446NY
REF. FAA STC: SR01446NY

Field experience has shown that the forward D3199-1 brackets that are part of the D412-698-011/-012/-017 Cabin Door Roller Kits and forward D3675-1 brackets that are part of the D412-698-031/-032/-033 Cabin Door Roller Kits may wear prematurely if the aircraft is flown a significant amount of time with the sliding doors open. Dart has developed the D412-698-027/-028 kits that includes a longer D3199-3/-4 bracket for installation in the first forward position of the D412-698-011/-017 kits to reduce wear on these forward brackets on aircraft equipped with aluminum doors. Likewise, Dart has developed the D412-698-031/-038 kits that includes a longer D3675-5/-6 bracket for installation in the first forward position of the D412-698-031/-032/-033 kits to reduce wear on these forward brackets on aircraft equipped with composite doors.

The D412-698-027/-028/-037/-038 kits may be installed as follows:

1) Remove the sliding door from the aircraft per Section 52.1 of ICA-D412-698

2) Remove the first forward D3199-1 or D3675-1 brackets as applicable.

3) Install the D3199-3/-4 or D3675-5/-6 bracket as applicable in lieu of the removed -1 bracket. Refer to Figure 1 Sheet 2 (aluminum doors) or Figure 2 Sheet 3 (composite doors).

4) Torque hardware per ICA-D412-698.

5) Re-install sliding door on aircraft per Section 52.2/52.3 of ICA-D412-698 as applicable.

6) Make entry in technical records of aircraft indicating that DSI 9484 has been completed.

PARTS LIST

UNCONTROLLED COPY SUBJECT TO AMENDMENT WITHOUT NOTICE WORK ORDER 85380 MLJ 12/06/05

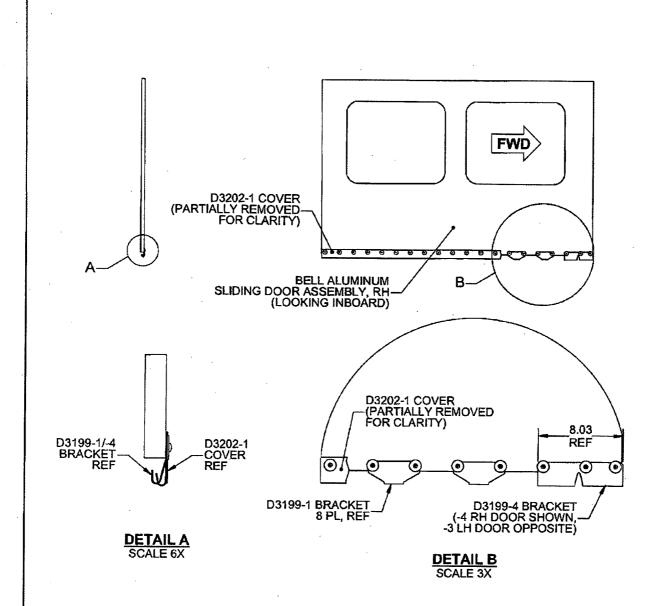
= 4					, in					
<b>+</b>	DESCRIPTION	BRACKET KIT, LH (ALUMINUM DOORS)	BRACKET KIT, RH (ALUMINUM DOORS)	BRACKET KIT, LH (COMPOSITE DOORS)	BRACKET KIT, RH (COMPOSITE DOORS)	FORWARD BRACKET, LH	FORWARD BRACKET, RH	FORWARD BRACKET, LH	FORWARD BRACKET, RH	
	PART NUMBER	D412-698-027	D412-698-028	D412-698-037	D412-698-038	D3199-3	D3199-4	D3675-5	D3675-6	
	QTY. -038				×				1	
	QTY. -037			×				1	•	
	31Y QTY. QTY. QTY.		×				-			
	Q:TY -027	×				-				

# WEIGHT AND BALANCE

Applicability		Aircraft Equipped with	P/N 205-032-669-XXX Aluminum Doors	Aircraft Equipped with P/N 412-030-300-XXX Composite Doors		
Longitudinal m Moment		16.37 in-lb 0.18 m-kg	16.37 in-lb 0.18 m-kg	18.41 in-lb 0.21 m-kg	18.41 in-lb 0.21 m-kg	
Long	Arm	102.3 in 2.60 m	102.3 in 2.60 m	102.3 in 2.60 m	102.3 in 2.60 m	
Lateral	Moment	-7.68 in-lb -0.09 m-kg	+7.68 in-lb +0.09 m-kg	-8.64 in-lb -0.10 m-kg	+8.64 in-lb +0.10 m-kg	
ב	Arm	-48.0 in -1.22 m	+48.0 +1.22 m	-48.0 in -1.22 m	+48.0 in +1.22 m	
Weight	)	0.16 lb 0.07 kg	0.16 lbs 0.07 kg	0.18 lb 0.08 kg	0.18 lb 0.08 kg	
Installation		D412-698-027 Bracket Kit	D412-698-028 Bracket Kit	D412-698-037 Bracket Kit	D412-698-038 Bracket Kit	

CANADA DEPARTMENT OF TRANSPORT AIRCRAFT CERTIFICATION BRANCH DAO # 01-0-01	BY: APPROVED b. SHEPHERD (DE#02)	DATE: 11.03.21 CERT, NO.: SH92-43 ISSUE NO.: 5

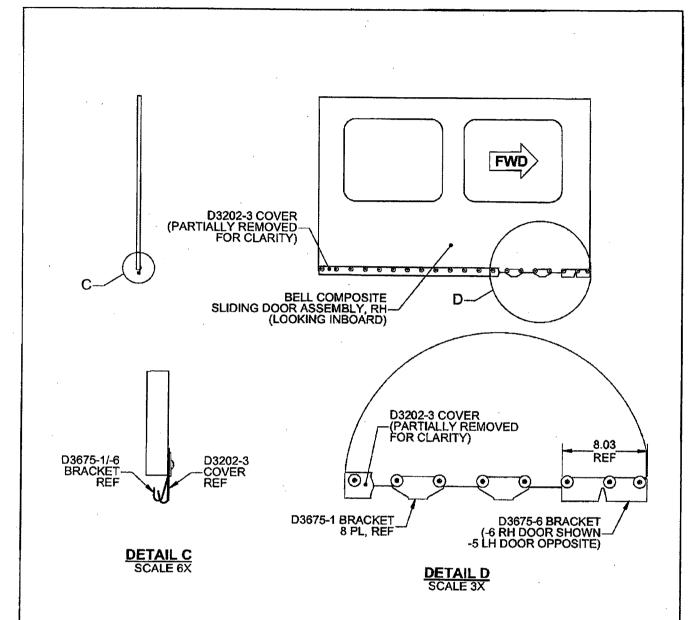
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8	ADDED TITLE A	ADDED -037/-038 KIT FOR TITLE ACCORDINGLY.	ADDED -037/-038 KIT FOR COMPOSITE DOORS; UPDATED TITLE ACCORDINGLY.	WB.	10.12.17	
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# FIGURE 1: D3199-3/-4 BRACKET INSTALLATION ON ALUMINUM DOORS (-4 SHOWN, -3 OPPOSITE)



DESIGN	8	DART AEROSPACE LTD	
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# FIGURE 2: D3675-5/-6 BRACKET INSTALLATION ON COMPOSITE DOORS (-6 SHOWN, -5 OPPOSITE)



DESIGN 8		DART AEROSPACE LTD				
DRAWN	13	HAWKESBURY, ONTARIO, CANADA				
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DATE 11.0	03.21	COPYRIGHT © 2009 BY DART AEROSPACE LTD  THIS DOQUERT IS PRIVATE AND CORPORATUL AND IS EXPRESD ON THE SUPPLESS CONDITION THAT IT IS  NOT TO BE USED FOR ANY PURPOSE OR COPIED OR COMMUNICATION TO NOT OTHER PERSON WITHOUT  WRITTEN PERSONS FOR INDIANY AEROSPACE LTD.  WRITTEN PERSONS FOR INDIANY AEROSPACE LTD.				

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## DART SERVICE INSTRUCTION

TO AMEND INSTALLATION INSTRUCTIONS IIN-D412-698 REV. E OR LATER APPROVED REVISION AND ICA-D412-698 REV. 5 OR LATER APPROVED REVISION

REF. TCCA STC: SH92-43 REF. FAA STC: SR01446NY REF, EASA.IM.R.S.00709

Field experience has shown that the forward D3199-1 brackets that are part of the D412-698-011/-012/-017 Cabin Door Roller Kits and forward D3675-1 brackets that are part of the D412-698-031/-032/-033 Cabin Door Roller Kits may wear prematurely if the aircraft is flown a significant amount of time with the sliding doors open. Dart has developed the D412-698-027/-028 kits that includes a longer D3199-3/-4 bracket for installation in the first forward position of the D412-698-011/-012/-017 kits to reduce wear on these forward brackets on aircraft equipped with aluminum doors. Likewise, Dart has developed the D412-698-037/-038 kits that includes a longer D3675-5/-6 bracket for installation in the first forward position of the D412-698-031/-032/-033 kits to reduce wear on these forward brackets on aircraft equipped with composite doors. equipped with composite doors.

The D412-698-027/-028/-037/-038 kits may be installed as follows:

1) Remove the sliding door from the aircraft per Section 52.1 of ICA-D412-698

2) Remove the first forward D3199-1 or D3675-1 brackets as applicable.

3) Install the D3199-3/-4 or D3675-5/-6 bracket as applicable in lieu of the removed -1 bracket. Refer to Figure 1 Sheet 2 (aluminum doors) or Figure 2 Sheet 3 (composite doors).

4) Torque hardware per ICA-D412-698.

5) Re-install sliding door on aircraft per Section 52.2/52.3 of ICA-D412-698 as applicable.

6) Make entry in technical records of aircraft indicating that DSI 9484 has been completed.

### **PARTS LIST**

QTY -027	QTY. -028	QTY:	QTY. -038	PART NUMBER	DESCRIPTION
X			-	D412-698-027	BRACKET KIT, LH (ALUMINUM DOORS)
	X			D412-698-028	BRACKET KIT, RH (ALUMINUM DOORS)
		X		D412-698-037	BRACKET KIT, LH (COMPOSITE DOORS)
J			Х	D412-698-038	BRACKET KIT, RH (COMPOSITE DOORS)
1				D3199-3	FORWARD BRACKET, LH
	1			D3199-4	FORWARD BRACKET, RH
	1	-1:		D3675-5 <i>■7</i>	FORWARD BRACKET, LH
			-1-	D3675-6	FORWARD BRACKET, RH

### **WEIGHT AND BALANCE**

Installation	Weight	Lateral		Longitudinal		Applicability	
		Arm	Moment	Arm	Moment		
D412-698-027 Bracket Kit	0.16 lb 0.07 kg	-48.0 in -1.22 m	-7.68 in-lb -0.09 m-kg	102.3 in 2.60 m	16.37 in-lb 0.18 m-kg	Aircraft Equipped with P/N 205-032-669-XXX Aluminum Doors	
D412-698-028 Bracket Kit	0.16 lbs 0.07 kg	+48.0 +1.22 m	+7.68 in-lb +0.09 m-kg	102.3 in 2.60 m	16.37 in-lb 0.18 m-kg		
D412-698-037 Bracket Kit	0.18 lb 0.08 kg	-48.0 in -1.22 m	-8.64 in-lb -0.10 m-kg	102.3 in 2.60 m	18.41 in-lb 0.21 m-kg	Aircraft Equipped with	
D412-698-038 Bracket Kit	0.18 lb 0.08 kg	+48.0 in +1.22 m	+8.64 in-lb +0.10 m-kg	102.3 in 2.60 m	18.41 in-lb 0.21 m-kg	Composite Doors	

CANADA DEPARTMENT OF TRANSPORT AIRCRAFT CERTIFICATION BRANCH DAO # 01-O-01

PPROVED ERD (DE # 02)

DATE: CERT. NO.: ISSUE NO .:

<u>11.03.21</u> SH92-43

C	UPDATE	ED -027/-028/-0	MB	11.03.21			
В	ADDED -037/-038 KIT FOR COMPOSITE DOORS; UPDATED TITLE ACCORDINGLY.				10.12.17		
A	NEW ISSUE				09.10.16		
REV.	DESCRIPTION				DATE		
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DATE 11.03.21			COPYRIGHT © 2009 BY DART AEROSPACE LTD.  THE DOCUMENT IS PRIVATE AND OBJECTORIAL AND IS SUPPLIED ON THE EXPRESS CONSTITION THAT IT IS  NOT TO BE USED FOR ANY PURPOSE OR COMED OR COMMUNICATION ONLY OTHER PERSON WITHOUT				